

Serial No. Not Yet Assigned
Atty. Doc. No. 2003P15370WOUS
Amendments To The Specification:

In the English translation document, please delete the term --Description-- at page 1, line 1.

In the English translation document, please add the paragraph at page 1, line 5, after the title, as follows:

--CROSS REFERENCE TO RELATED APPLICATIONS

This application is the US National Stage of International Application No. PCT/EP2005/000925, filed January 31, 2005 and claims the benefit thereof. The International Application claims the benefits of European Patent application No. 04003669.1 filed February 18, 2004. All of the applications are incorporated by reference herein in their entirety.--

In the English translation document, please add the section heading at page 1, line 5, after the newly added CROSS REFERENCE TO RELATED APPLICATIONS section with the new section heading, as follows:

--FIELD OF THE INVENTION--

In the English translation document, please insert the section heading at page 1, line 14, as follows:

--BACKGROUND OF THE INVENTION--

In the English translation document, please add the paragraph at page 2, line 7, as follows:

--Furthermore, US 5,154,578 discloses a compressor housing of an aircraft gas turbine, in which the radial supports connecting an external housing and an internal housing of the compressor may have a heating or cooling medium flow through them to set the radial gap of the rotor blades of the compressor.--

In the English translation document, please add the paragraph at page 2, before line 8, as follows:

--In addition, US 5,605,437 discloses a device situated in the compressor housing for

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reducing the oscillations of the radial gap of freestanding guide blades of the compressor. The guide blade rings each have a base ring channel, i.e., situated in the compressor housing, for this purpose, which a heating medium may flow through. The ring channels are connected to one another by overflow channels, so that the heating medium may flow through the ring channels sequentially independently of the compressor operation.--

In the English translation document, please insert the section heading at page 2, before line 8, after the newly added paragraph, as follows:

--SUMMARY OF THE INVENTION--

In the English translation document, please amend the paragraph at page 2, line 19, as follows:

--This object is achieved according to the present invention by the features of ~~claim 1~~ the claims. For this purpose, in a gas turbine having a turbine and a compressor comprising a compressor housing, the compressor being tapped to cool the turbine using at least one tap line to remove compressed or partially compressed air, the tap line has a locking device, in particularly a valve. The same object is also achieved by a compressor or a compressor housing having the features of ~~claim 6~~ the claims.--

In the English translation document, please amend the paragraph at page 2, line 29, as follows:

--Furthermore, the object is achieved by a method for operating such a gas turbine having the features of ~~claim 4~~ the claims, in which the locking device, particularly the valve, is closed or partially closed as the gas turbine is shut down.--

In the English translation document, please insert the section heading at page 5, line 4, as follows:

--BRIEF DESCRIPTION OF THE DRAWINGS--

In the English translation document, please insert the section heading at page 5, line 22, as follows:

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--DETAILED DESCRIPTION OF THE INVENTION--